

# RECENT ACTIVITIES OF THE FP7-ESP<sub>a</sub>CE CONSORTIUM

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**ABSTRACT.** The consortium ESP<sub>a</sub>CE (European Satellite Partnership for Computing Ephemerides) is composed of seven European institutes: IMCCE ((Institut de Mécanique Céleste et de Calcul des Ephémérides, Paris Obs.), ROB (Royal Observatory of Belgium), TUB (Technical University of Berlin), JIVE (Joint Institute for VLBI in Europe), TUD (Delft University of Technology), French space agency (CNES) in France and German Aerospace Center (DLR) in Germany. The objective of this FP7 European project is to provide new accurate ephemerides of natural satellites and spacecraft. For this goal many astrometric data issued from ground-based observations as well as from space observations have been analyzed and reduced. On the other hand new technologies applied to the positioning of spacecraft are also studied. The ESP<sub>a</sub>CE project addresses also data related to gravity and shape modeling, control point network and rotational parameters of natural satellites. The accuracy improvement of these ephemerides makes them a powerful tool for the analysis of space missions or the preparation of future missions, or for the determination of some physical parameters.

## 1. THE CONCEPT AND OBJECTIVES

The first purpose of the ESP<sub>a</sub>CE project is to explore and understand the physical phenomena, the physical structure and the dynamical processes of the Earth's Moon, the Martian moons Phobos and Deimos, as well as the numerous planetary moons of the Solar system, and of Jupiter and Saturn in particular, their origin, their dynamics and their evolution. The second purpose of this project is to link celestial mechanics, dynamics, space science, Radio Science, LLR (Lunar Laser Ranging), VLBI (Very Long Baseline Interferometry) and astrometry. The main outputs are the improvement of the ephemerides of the Martian, Jovian, Saturnian and Uranian satellites and of orbits of spacecraft, as well as constraints on the interior and dissipation processes associated with these planetary systems.

The main part of the activity is focused on the extraction and analysis of astrometric data from spacecraft measurements that have not yet been used in the orbit dynamic reconstruction and on the combination of these data with ground-based astrometric data. The project will also advance the European expertise in ultra-precise tracking of planetary probes. By these means, we intend to provide new dynamical models for several natural satellites, a characterization of their rotation properties, and improve spacecraft orbit determination methods for space science.

This four year project, which began on 2011 June 1st, is organized in 12 work-packages: management of the project, Radio Science (RS), Laser Ranging (LR), VLBI, digitized data handling, astrometry, definition of coordination reference frames and improvement of planetary coordinate knowledge, methods for determination of spacecraft and satellites ephemerides, set up of databases, data access and distribution methods, educational and outreach activities, and scientific management (Thuillot et al., 2013).

The project concentrates at achieving maximum synergy between all the work packages above in order to deliver to the professional communities and communities at large the best scientific products adequate

to the present day cutting edge space science and technology. For further information web pages are accessible at : <http://espace.oma.be/>

## 2. RECENT RESULTS

### *Astrometry from digitized photographic plates and telescopic observations*

The new technology ROB Digitizer (Damian) was used to digitally disclose with unprecedented precision the astrometric data archived in photographic plates of the Martian ( $\approx 500$  plates), the Galilean ( $\approx 550$  plates) and the Saturnian ( $\approx 1000$  plates) satellites mostly taken with the 26" refractor at USNO and at the South African Yale Station. Almost 200 plates from different locations will also be considered. Other telescopic observations, mutual events of planetary satellite systems (Uranus 2007, Jupiter 2009), have been reduced.

### *Astrometry from RS and Space images*

Precise determination of S/C orbit is done thanks to RS (MEX, Viking 1&2, Mariner 9 data) and will be used in a global astrometry solution of the satellite dynamical models. Besides, space imaging is used (Pasewaldt et al., 2012). 239 Phobos and 136 Deimos images have been studied and lead to observational accuracies between 0.5 and 3.6 km. Intersatellite measurements of the least distance during Martian moons encounters show also promising astrometry results.

### *Astrometry from LR and VLBI tracking*

New technologies for providing high accurate astrometry of S/C through tracking or ranging data are studied. The combination of RS and VLBI observations during a flyby of spacecraft around a planet or moons is under study (Duev et al., 2012). Besides, we foresee to contribute to the positioning of Gaia using VLBI tracking techniques. Results using 1-Way LR to LRO have been obtained and show rms of 13.2cm. This may let us expecting a performance  $e < 1$  cm with a dedicated two-way system. Performance study of Phobos laser ranging concept for geodetic observables is currently in progress.

### *Shape modeling, reference system and rotation*

ESPaCE intends to provide coordinate systems of several satellites and rotation parameters (Rambaux et al., 2012). A Phobos control point network is under development. A provisional version has been used for providing space astrometry but also for providing measures of rotation parameters. Potential applications to the libration study of the Saturnian and Jovian satellites are foreseen. The development of reference shape and reference system of icy satellites is also under study. Interior and dynamical parameters will be inferred.

### *New satellite and spacecraft ephemerides*

New ephemerides of the Martian and the Uranian moons have been developed. Ground-based astrometry and space astrometry (MEX, Phobos2, Viking data) are used for Phobos and Deimos. Post-fit residuals show typical accuracy of 65 mas ( $\approx 20$  km). For the Uranian moons, residuals of micrometer observations, photographic plates and CCD frames covering 1870-2012 are obtained. Mutual events residuals of these satellites show very accurate results with rms of 20 mas (Arlot et al., 2013). The ephemerides are accessible at <ftp.imcce.fr>.

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## 3. REFERENCES

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