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brahim.lamine@upmc.fr http://www.lkb.ens.fr Framework of metric extension of GR



- Gravitation ↔ geometry of space-time
 - \rightarrow equivalence principle preserved
 - \rightarrow metric theory $ds^2 = g_{\mu\nu} dx^{\mu} dx^{\nu}$
- Coupling of curvature to energy may be modified.

$$\mathbf{E}_{\mu\nu} \equiv \mathbf{R}_{\mu\nu} - \frac{1}{2} g_{\mu\nu} \mathbf{R} = \frac{8\pi \mathbf{G}}{c^4} \mathbf{T}_{\mu\nu} \longrightarrow \mathbf{E}_{\mu\nu}[k] = \chi_{\mu\nu}^{\ \rho\sigma}[k] \mathbf{T}_{\rho\sigma}[k]$$

 \rightarrow more general linear relation in Fourier space.

- Two main consequences :
 - \rightarrow different couplings for the two independent sectors of traceless and traced E_{µv}.
 - \rightarrow G is replaced by two "running constants" G⁽⁰⁾[k] and G⁽¹⁾[k].
- It defines two sectors.
 - → special combinations of these two sectors can either mainly affect matter or light propagation.

Framework in a simple solar system model

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• Solution of Einstein equation in isotropic gauge $ds^2 = g_{00}$

$$f^{2} = g_{00} \left(c \mathrm{d}t \right)^{2} + g_{rr} \, \mathrm{d}t$$

$$g_{00} = 1 + 2\phi + 2\phi^2 + \dots$$
, $-g_{rr} = 1 - 2\phi + \dots$ $\phi = -\frac{GM}{rc^2}$

PPN framework

 $g_{00} = 1 + 2\phi + 2\beta\phi^2 + \dots , \quad -g_{rr} = 1 - 2\gamma\phi + \dots$

 \rightarrow 30 years of PPN tests have selected a vicinity of GR in the PPN framework.

 $\gamma - 1 < 2 \times 10^{-5}$, $\beta - 1 < 10^{-4}$

 The two sectors of the new framework matches a liberty in the two functions g₀₀ and g_{rr}

$$g_{00} = [g_{00}]_{\text{GR}} + \delta g_{00}$$
$$g_{rr} = [g_{rr}]_{\text{GR}} + \delta g_{rr}$$

Example of explicit phenomenological model

• We can search for Taylor expansion valid inside the solar system.

$$\delta g_{00}=2\sum_{n>0}lpha_n r^n \quad,\quad \delta g_{rr}=2\sum_{n>0}\chi_n r^n$$

 \rightarrow equivalently, it can also be analyzed as PPN parameters depending on distance.

$$\beta(r) - 1 = \left(\frac{c^2}{GM}\right)^2 \sum_{n>0} \alpha_n r^{n+2} \quad , \quad \gamma(r) - 1 = -\frac{c^2}{GM} \sum_{n>0} \chi_n r^{n+1}$$

- Modifications of g₀₀ are already well constrained by planetary tests.
 - \rightarrow in particular, a modification of g₀₀ only (sector 1) accounting for the Pioneer anomaly would have been seen.
- Example of phenomenological modification of g_{rr} : $\delta g_{rr} = 2\chi_2 r^2$ \rightarrow it corresponds to a constant curvature (outside the sources !) $R_{rr} = 8\chi$

Effects on matter

• Geodesic equation

$$\begin{aligned} \ddot{x}^{i}(t) &= \frac{c^{2}}{2} \eta^{ij} \partial_{j} h_{00} - c^{2} \eta^{ij} \partial_{0} h_{0j} + \frac{c}{2} \dot{x}^{i} \partial_{0} h_{00} - c \eta^{ik} \dot{x}^{j} \partial_{0} h_{kj} - c \eta^{ik} \dot{x}^{j} \left(\partial_{j} h_{0k} - \partial_{k} h_{0j} \right) \\ &+ \dot{x}^{i} \dot{x}^{k} \partial_{k} h_{00} - \frac{\dot{x}^{j} \dot{x}^{k} \eta^{il} \left(\partial_{j} h_{kl} - \frac{1}{2} \partial_{l} h_{jk} \right) - \frac{\dot{x}^{i} \dot{x}^{k} \dot{x}^{l}}{c} \left(\frac{1}{2} \partial_{0} h_{kl} - \partial_{k} h_{0l} \right) \end{aligned}$$

 \rightarrow anomalous (coordinate) acceleration

$$egin{aligned} & [\delta a_r]_{\chi_2} = 2\chi_2\,r(v_r^2-v_ heta^2) \ & [\delta a_ heta]_{\chi_2} = 4\chi_2\,rv_rv_ heta \end{aligned}$$

- → the radial acceleration is a negative constant (if χ_2 >0) for bounded trajectories and a positive constant (if χ_2 >0) for escaped trajectories.
- \rightarrow the orthoradial acceleration vanishes for circular orbit and radial escape.
- The anomalous acceleration has to stay small to remain compatible with planetary ephemerides.

$$\delta a \sim \chi_2 \text{GM}$$

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Effects on spacecraft Doppler traking

Effect on light propagation

 \rightarrow modification of Shapiro time delay

$$c\delta \mathcal{T} = -rac{1}{3} \, \chi_2 \mathrm{R}_{12} \left(\mathrm{R}_{12}^2 - 3 \mathrm{R}_{12} \sqrt{r_1^2 - b^2} + 3 r_1^2
ight)$$

→ Doppler observable and overall optical range - rate $y = \frac{\delta \nu}{\nu} \equiv \frac{1}{c} \frac{d\ell}{dt}$

 \rightarrow effective acceleration along the line of sight

$$a\sim \frac{c}{2}\dot{y}$$



- Consequences :
 - → for non radial motion, the contribution coming from light is bigger than the one coming from the modified spacecraft dynamics.
 - → The effect on light naturally leads to periodic anomalies modulated by the annual and diurnal motions of the Earth.

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Conclusion

• There are two sectors in an extended metric extension of GR.

- The second sector could have escaped planetary tests but could still produce observable effects in Doppler tracking of spacecrafts.
 - \rightarrow Continue test of gravity with spacecraft tracking !

- The second sector produces periodic anomalies modulated by the annual and diurnal motions of the Earth :
- \rightarrow through the range or Doppler observable for spacecraft tracking.
- \rightarrow through anomalies of declinaison and longitude in observations of planets.

$$\delta \boldsymbol{n} = \mathbf{grad} \ c \delta \mathcal{T}$$

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