

INPOP06: the new European Planetary, Moon and Earth rotation Ephemeris

A. Fienga⁽¹⁾, J.Laskar⁽¹⁾, H.Manche⁽¹⁾, M.Gastineau⁽¹⁾,
C. Le Poncin-Lafitte^{(1),(2)}

(1)IMCCE / Observatoire de Paris
(2)Lohrmann Observatory, Dresden

INPOP Planetary ephemeris

INPOP06
(Fienga et al. 2006)

- Intégrateur Numérique Planétaire de l'Observatoire de Paris
- began in 2003 with J.Laskar et al.

- Numerical integration with extended precision 80b
- Planets, Asteroids, Moon, Earth Rotation
- Einstein-Infeld-Hoffman equation of motion
- INPOP(TCB) and INPOP(TDB)
- Fit to ~ 50 000 space and Earth-based data
- Asteroid masses and densities
- J2 estimated (β, γ tested)
- INPOP realisation of TCB

- ESA ephemeris (GAIA)
- ESA tracking data (MEX)

INPOP General Aspect – I: Planets and Asteroids

$$\ddot{\mathbf{r}}_i = \sum_{j \neq i} \frac{\mu_j (\mathbf{r}_j - \mathbf{r}_i)}{r_{ij}^3} \left\{ 1 - \frac{2(\beta + \gamma)}{c^2} \sum_{l \neq i} \frac{\mu_l}{r_{il}} - \frac{2\beta - 1}{c^2} \sum_{k \neq j} \frac{\mu_k}{r_{jk}} \right. \\ \left. + \gamma \left(\frac{\dot{s}_i}{c} \right)^2 + (1 + \gamma) \left(\frac{\dot{s}_j}{c} \right)^2 - \frac{2(1 + \gamma)}{c^2} \dot{\mathbf{r}}_i \cdot \dot{\mathbf{r}}_j \right. \\ \left. - \frac{3}{2c^2} \left[\frac{(\mathbf{r}_i - \mathbf{r}_j) \cdot \dot{\mathbf{r}}_j}{r_{ij}} \right]^2 + \frac{1}{2c^2} (\mathbf{r}_j - \mathbf{r}_i) \cdot \ddot{\mathbf{r}}_j \right\} \\ + \frac{1}{c^2} \sum_{j \neq i} \frac{\mu_j}{r_{ij}^3} \left\{ [\mathbf{r}_i - \mathbf{r}_j] \cdot [(2 + 2\gamma) \dot{\mathbf{r}}_i - (1 + 2\gamma) \dot{\mathbf{r}}_j] \right\} (\dot{\mathbf{r}}_i - \dot{\mathbf{r}}_j) \\ + \frac{3 + 4\gamma}{2c^2} \sum_{j \neq i} \frac{\mu_j \ddot{\mathbf{r}}_j}{r_{ij}} \quad (\text{Newhall, 1983})$$

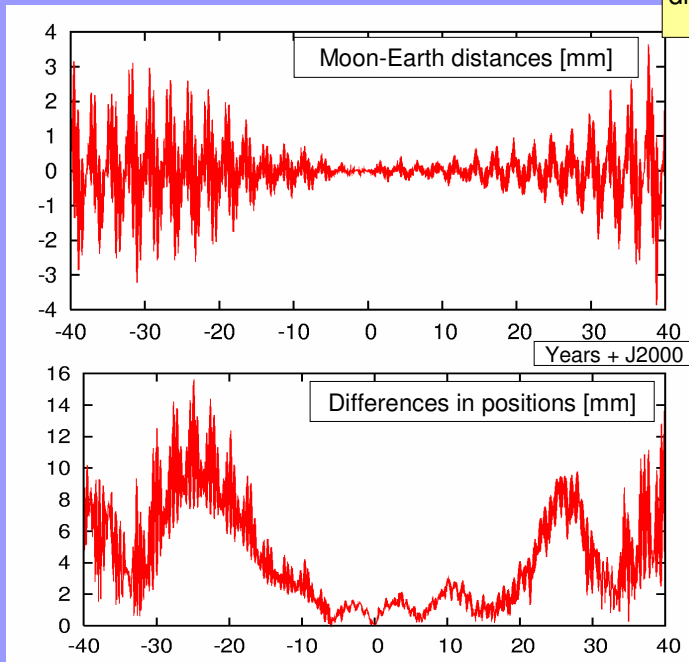
- Einstein-Infeld-Hoffman equation of motion
- BCRF (IAU 2001)
- β, γ, J_2
- Asteroids integrated like planets
- Asteroid Ring perturbations included
- SSB

INPOP General Aspect – II: The Moon and Moon Libration

- EIF perturbations of planets, Sun and 300 asteroids (integrated)
- **Non-spherical bodies** \Leftrightarrow **point mass bodies**
 - Sun (J2) \Leftrightarrow Planets
 - Earth (J2) \Leftrightarrow (Moon, Sun, Venus, Jupiter)
 - Moon (J2, C21, C22, S21, S22) \Leftrightarrow (Earth, Sun, Venus, Jupiter)
- **Deformation of extended bodies (tides)** \Leftrightarrow **point mass bodies**
 - Earth (Sun, Moon) \Leftrightarrow (Moon, Sun, Venus, Jupiter)
 - Moon (Spin, Earth, Sun) \Leftrightarrow (Earth, Sun, Venus, Jupiter)
- Earth Shape \Leftrightarrow Moon shape (torque exerted by the Moon)
- DE/LE405 : fitted to LLR / INPOP06: Not yet fit to LLR data (in progress)

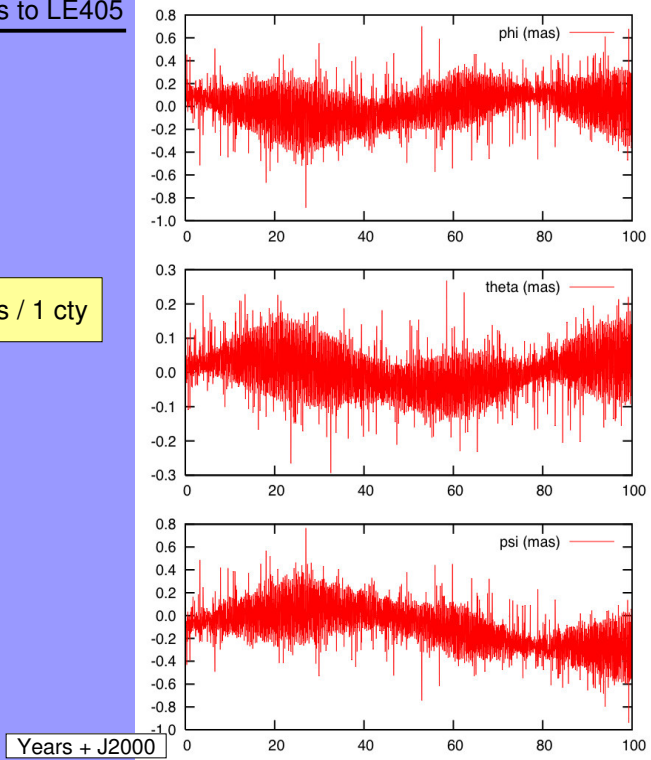
The Moon Positions: Comparisons to LE405

Few mm for Moon geocentric distances on LLR data interval



The Moon Libration: Comparisons to LE405

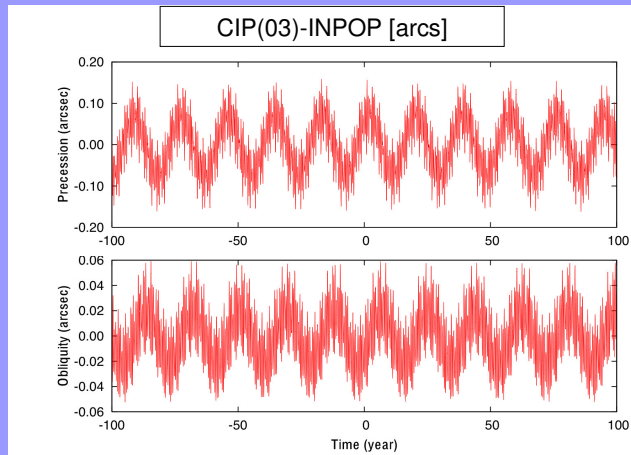
< 1 mas for the 3 Euler angles / 1 cty



INPOP General Aspect – III: Earth rotation

See Jacques Laskar Presentation ...

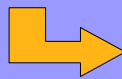
- INPOP Earth Rotation solution = Integration of the angular momentum
- Takes into account the Earth tidal deformation
- does not depend of the Inner Earth modelling



General Relativity and INPOP

For ESA missions (GAIA),

- Einstein-Infeld-Hoffman equation of motion
- Barycentric Celestial reference frame : IAU 2001 BCRS if $T_{\text{eph}} = \text{TCB}$
- Each planetary ephemeris realises its own TCB (Klioner 05)



$(\text{TCB}_{\text{inpop}} - \text{TCG})$ estimated via an iterative process and
INPOP fit to observations

(Fienga, Le Poncin-Lafitte et al., 2006)

- In INPOP, RG SSB definition is

$$\sum_i \mu_i^* r_i = 0 \text{ and } \sum_i \mu_i^* dr_i / dt + d\mu_i^* dt r_i = 0$$

Work in progress...

INPOP06 Data Set

- [1898-2006] data set
- JPL space data
- US/Russian direct radar
- Transit data
- Original CCD data
- **ESA MEX/VEX**

**JPL/Observatoire de Paris
Astrometric Planetary Data
Base**

Planets	Types of Observations	Time Interval	SOURCES	< σ >	
Mercury	Radar	1972-1997	Goldstone,Arecibo,Evpatoria	1 km	443
Venus	Radar	1970-1990	Goldstone,Arecibo,Evpatoria	1 km	511
	Spacecraft VLBI	1991-1994	Magellan	1 mas	18
	<i>Spacecraft radar</i>	<i>Sept 2006</i>	ESA VEX in sept 06		
Mars	Spacecraft radar	1978-2005	JPL Vkg,Pathf,MGS,MO	5/10 m	10474
	Spacecraft radar	2005-2006	ESA MEX	2 m	19000
	Spacecraft VLBI	2001-2003	JPL MGS,MO	1 mas	44
	Spacecraft Doppler	1980, 1999	JPL Vkg, Pathf	1 km	3020
Jupiter	Spacecraft VLBI	1996-1997	JPL Galileo	1/10 mm/s	24
	CCD or transit (α,δ)	1970-2004	USNO,Nikolaev,CAMC,Bdx, MPC	500 mas	3191
Saturne	CCD or transit (α,δ)	1970-2004	USNO,Nikolaev,CAMC,Bdx, MPC	500 mas	3863
Uranus	CCD or transit (α,δ)	1900-2004	USNO,Nikolaev,CAMC,Bdx, MPC	500 mas	3849
Neptune	CCD or transit (α,δ)	1900-2004	USNO,Nikolaev,CAMC,Bdx, MPC	500 mas	3898
Pluto	CCD or transit (α,δ)	1988-2004	USNO,Nikolaev,CAMC,Bdx, MPC	500 mas	1124

INPOP06 adjusted physical parameters

Planet Initial Conditions +

	Unit	DE405 (Standish, 98)	DE411 (Standish, 04)	EMP04 (Pitjeva 05)	DE414 (Standish, 05)	INPOP06 (Fienga, et al 06)
Mass of Ceres	$10^{-10} M_{\oplus}$	4.64	4.554	4.753±0.007	4.699	4.746±0.006
Mass of Vesta	$10^{-10} M_{\oplus}$	1.34	1.280	1.344±0.001	1.358	1.338±0.002
Mass of Pallas	$10^{-10} M_{\oplus}$	1.05	1.059	1.027±0.003	1.026	0.995±0.003
Mass of Iris	$10^{-10} M_{\oplus}$			0.063±0.001	0.060	0.089±0.002
Mass of Bambergia	$10^{-10} M_{\oplus}$			0.055±0.001	0.047	0.060±0.002
Density of C class		1.8	1.8	1.4	1.6 ±0.22	1.93±0.12
Density of S class		2.4	2.4	3.5	2.07	2.13±0.11
Density of M class		5.0	5.0	4.5	4.3±0.43	4.47±0.012
Mass of Asteroid ring	$10^{-10} M_{\oplus}$			3.55±0.35	0.329	0.34±0.15
Distance of ring	UA			3.13±0.05	2.8	2.8
Sun J2	$X 10^{-7}$	2	2	1.9±0.3	2.3±2.5	1.95±0.55

INPOP06b: $1 \leq \beta \leq 1.0001$, $1 \leq \gamma \leq 1.00002$

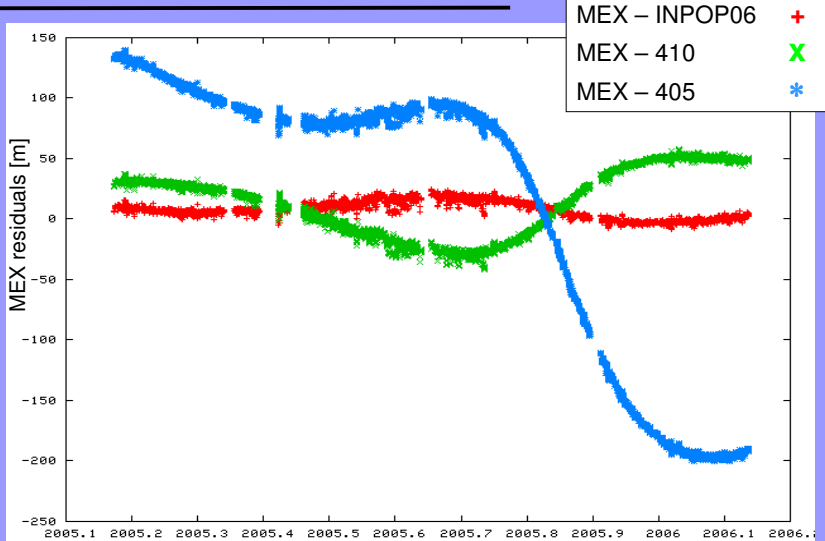
INPOP06 / DE414

- INPOP06: Original dynamical models
Asteroids / **all planets**
5 Bigs + 297 in 3 classes+ Asteroid ring
General Ephemeris over [1899:2005.43]

- DE414 : **64 Asteroids GM fitted + Asteroid ring**
236 in 3 classes
Mars & Saturn Space missions dedicated
[2000:2005.3]

Planets	Types of Observations		DE414	INPOP06
Mercury	Radar [m]	444	-596 ± 956	-108 ± 871
Venus	Radar [m]	511	-3578.0 ± 4671	-1126.0 ± 4527
	Spacecraft VLBI [mas]	18	1.7 ± 2	0.6 ± 2
Mars	Vkg radar [m]	1256	-5.7 ± 18	0.2 ± 20
	MGS Radar [m]	10474	7.4 ± 4.1	2.5 ± 7.5
	<i>MEX Radar [m]</i>	<i>19000</i>	<i>11.2 ± 12.1</i>	<i>8.3 ± 6.7</i>
	Spacecraft Doppler Vkg [mm/s]	1501	-0.26 ± 4.5	-0.25 ± 4.4
	Spacecraft Doppl MGS [mm/s]	1519	0.35 ± 1.0	-0.32 ± 1.2
	Spacecraft VLBI [mas]	44	0.04 ± 0.5	0.1 ± 0.5
Jupiter	Spacecraft VLBI [mas]	24	-1 ± 12	3 ± 12
	CCD or transit (α,δ) [mas]		(47 ± 222, 36 ± 198)	(23 ± 214, -26 ± 190)
Saturne	CCD or transit (α,δ) [mas]	444	(29 ± 280, -1 ± 196)	(-28 ± 270, -2 ± 196)
Uranus	CCD or transit (α,δ) [mas]	511	(11 ± 440, 13 ± 370)	(0.5 ± 351, 4 ± 361)
Neptune	CCD or transit (α,δ) [mas]	18	(12 ± 404, 11 ± 438)	(-0.4 ± 360, 0.5 ± 350)
Pluto	CCD or transit (α,δ) [mas]	1256	(13 ± 264, 8 ± 252)	(0.9 ± 250, -47 ± 190)

MEX (Morley 06) data




	Time interval	INPOP	DE410	DE405
MEX [m]	2005.18-2006.2	8.3 ± 6.7	14.3 ± 25.2	27.8 ± 112.3

Project


- **Lunar Laser Ranging and INPOP Lunar fit**

 with Observatoire de Paris (S.Bouquillon, J.Chapront)

- **Adjustement with the new VEX and MEX Data**

 with ESOC (T.Morlay)

- **Pulsar Timing for EMB orbit fit**

 • with Observatoire de Besançon and Nançay
(F.Vernotte, I.Cognard)
• Contact with ATNF (R. Edwards)

- **RG β , γ influences and VEX determinations**

 with Dresden Lohrmann Observatory
(C.Le Poncin-Lafitte)

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